

Measurements of laminar burning velocities and onset of cellular instabilities of methane-hydrogen-air flames at elevated pressures and temperatures

Erjiang Hu, Zuohua Huang*, Jiajia He, Jianjun Zheng, Haiyan Miao

State Key Laboratory of Multiphase Flow in Power Engineering, Xi'an Jiaotong University, 28 Xian Ning West Road, Xi'an 710049, Shaanxi Province, PR China

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ABSTRACT

An experimental study on laminar burning velocities and onset of cellular instabilities of the premixed methane-hydrogen-air flames was conducted in a constant volume combustion vessel at elevated pressures and temperatures. The unstretched laminar burning velocity and Markstein length were obtained over a wide range of hydrogen fractions. Besides, the effects of hydrogen addition, initial pressure and initial temperature on flame instabilities were analyzed. The results show that the unstretched flame propagation speed and the unstretched laminar burning velocity are increased with the increase of initial temperature and hydrogen fraction, and they are decreased with the increase of initial pressure. Early onset of cellular instability is presented and the critical radius and Markstein length are decreased with the increase of initial pressure. Flame instability is insensitive to initial temperature compared to initial pressure. With the increase of hydrogen fraction, significant decrease in critical radius and Markstein length is presented, indicating the increase in both diffusional-thermal and hydrodynamic instabilities as hydrogen fraction is increased.

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1. Introduction

With increasing concern on fossil fuel shortage and stringent emission regulations, the development of alternative fuel engines has attracted more and more attention around the world. Natural gas, is a clean fuel in which methane is its major component, is considered to be one of the most favorable fuels for engines, and the utilization of natural gas has been realized in the spark-ignition engines. However, due to the slow burning velocity of natural gas and its poor lean-burn capability, the natural gas spark-ignition engine still has its disadvantages like low thermal efficiency, large cycle-by-cycle variation, and poor lean-burn capability, and these will decrease the engine power output and increase the fuel consumption [1,2]. One of the effective methods to solve the problem of slow burning velocity of natural gas is to mix natural gas with a fuel that possesses high burning velocity. Hydrogen is regarded as the best gaseous candidate due to its high burning velocity, and the combination of natural gas with hydrogen is expected to improve the lean-burn characteristics and decrease the engine emissions [3–8].

In order to understand the effect of hydrogen addition into natural gas on engine combustion, the understanding of

 ^{*} Corresponding author. Tel.: +86 29 82665075; fax: +86 29 82668789.
 E-mail address: zhhuang@mail.xjtu.edu.cn (Z. Huang).

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fundamental combustion behavior with hydrogen enriched methane is indispensable. It will provide an in-depth understanding of combustion. Laminar burning velocity is the fundamental parameter in combustion and it is also the base parameter for turbulent burning velocity determination. Laminar burning velocity can be used to validate the chemical reaction mechanisms [9,10] and is of practical importance in the design and optimization of internal combustion engines and power plant burners [11]. There are three approaches to measure the laminar burning velocity, they are, the stagnation plane flame method [12,13], the heat flux method [14,15] and the combustion bomb method [16,17]. The stagnation plane flame method can establish different flame configurations, but it is difficult to draw a clear flame front and to stabilize the flame under the high-pressure conditions. With respect to the heat flux method, further work should be done to determine the heat loss as a function of inlet velocity and to extrapolate the results to zero heat loss to get the adiabatic burning velocity. The combustion bomb method utilizes the prototypical propagating spherical flame configuration and has drawn the particular attention due to its simple flame configuration, well-defined flame stretch rate and well-controlled experimentation [18,19]. Practically, the outwardly propagating flame is more similar to the flame propagation in the spark-ignition engines. In this study, the laminar burning velocities of methanehydrogen-air mixtures were measured by using the spherically expanding flame.

Many previous researches on laminar burning velocities concentrated on the methane-air flames [13,16,18,20,21] and/ or the hydrogen-air flames [9,19,22-26]. Recently, some experimental studies reported the measurement of laminar burning velocity for the methane-hydrogen-air flames [11,13,27-33]. Liu et al. [29] and Huang et al. [30] conducted more extensive experimental studies over a wide range of equivalence ratios and hydrogen fractions in the methanehydrogen-air flames. Yu et al. [13] studied the laminar burning characteristics of methane-hydrogen-air flames with the assumption that the stoichiometrically small amounts of hydrogen in the mixture were completely consumed and found a linear correlation between laminar burning velocity and hydrogen fraction. Law and Kwon [27] studied the potential of hydrocarbon addition into hydrogen to suppress explosion hazards and found that a small or moderate amount of methane addition could remarkably reduce the laminar burning velocities and would suppress the propensity of onset of both diffusional-thermal instability and hydrodynamic cellular instability in hydrogen-air flames. However, these studies were mostly conducted at the atmospheric pressure and at the room temperature, and a few literatures on combustion of hydrogen enriched methane-air flames at elevated pressures and temperatures were reported.

As lean flames are likely to be unstable in both methaneair mixtures and hydrogen-air mixtures, lean combustion is widely used in engineering fields. Thus, this study will select the equivalence ratio of 0.8 to study the basic characteristics of the flames. The outwardly propagating spherical flame was used to obtain the laminar burning parameters, including the unstretched laminar burning velocity (u_1) and the burned gas Markstein length (L_b), which reflects the sensitivity of flame stability to flame stretch rate. In addition, the thermal expansion ratio (σ) and the laminar flame thickness (δ_l) of methane–hydrogen–air mixtures at elevated pressures and at different hydrogen fractions were provided to analyze the flame stability. Critical flame radius, Peclet number and influence of diffusional-thermal and hydrodynamic instabilities on the transition to cellular flames were analyzed. This study will provide further information and is beneficial in understanding methane–hydrogen–air mixture flames.

2. Experimental setup and procedures

Fig. 1 shows the experimental setup. It includes a constant volume combustion chamber and the systems for heating, ignition, data acquisition and high-speed Schlieren photography. The combustion chamber is a cylindrical type with an inner diameter of 180 mm and volume of 5.5 L as shown in Fig. 2. The centrally located electrodes are used to ignite the premixed mixtures. Pressure transmitter, thermocouple, pressure transducer, inlet and outlet valves are mounted on the chamber body. Two guartz windows with 80 mm diameter are located at the two sides of the vessel to allow an optical accessiblity. A high-speed digital camera (HG-100K) operating at 10,000 frames per second is used to record the flame photos during flame propagation. Fuel and dry air are supplied into the chamber through the inlet valve corresponding to the given equivalence ratio and dilution ratio. Five minutes is awaited before starting the ignition to ensure the homogeneity of methane-hydrogenair mixtures.

In the experiments, the initial pressure is set at 0.1 MPa, 0.25 MPa, 0.5 MPa and 0.75 MPa and the initial temperature is set at 303 K, 373 K and 443 K, respectively. Hydrogen fractions are varied from 0% to 80%. Purities of methane and hydrogen in the study are 99.9% and 99.995%, respectively.

Hydrogen fraction (X_{H_2}) is defined as

$$X_{H_2} = \frac{V_{H_2}}{V_{CH_4} + V_{H_2}}$$
(1)

where V_{CH_4} and V_{H_2} are the volume fractions of methane and hydrogen in the fuel blends, respectively.

Total equivalence ratio (ϕ) is defined as

$$\phi = \frac{F/A}{(F/A)_{\rm st}} \tag{2}$$

where F/A is the fuel-air ratio and $(F/A)_{st}$ refers to the stoichiometric value of F/A.

Mixture can be expressed as

$$\left(1 - X_{H_2}\right)CH_4 + X_{H_2}H_2 + \left(\frac{2}{\phi}(1 - X_{H_2})\frac{X_{H_2}}{2\phi}\right)(O_2 + 3.762N_2)$$
(3)

There are three reactants in the mixture, so it is necessary to define the stoichiometric parameters for data reduction. In the present work, the mole fractions of methane, hydrogen, fuel blend and air are n_{CH_4} , n_{H_2} , n_{Fuel} , and n_{Air} , respectively.



Fig. 1 - Experimental setup.

Thus, the hydrogen and methane based equivalence ratios are defined as

$$\phi_{\rm H_2} = \frac{n_{\rm H_2}/n_{\rm Air}}{\left(n_{\rm Fuel}/n_{\rm Air}\right)_{\rm st}} \tag{4}$$

$$\phi_{\rm CH_4} = \frac{n_{\rm CH_4}/n_{\rm Air}}{(n_{\rm Fuel}/n_{\rm Air})_{\rm st}}$$
(5)

The denominator of ϕ_{H_2} and ϕ_{CH_4} is the stoichiometric fuel/ air molar ratio, and the numerator is the ratio of individual fuel to the amount of air available for its oxidation. Fig. 3 illustrates the equivalence ratio of hydrogen (ϕ_{H_2}) and methane (ϕ_{CH_4}) at different hydrogen percentages at the total equivalence ratio of 0.8. Since there are two fuels in the mixture, the Lewis number should be a weighed average value. In this study, as proposed by Law et al. [34], the Lewis number is evaluated from the following:

$$Le_{eff} = 1 + \frac{q_{H_2}(Le_{H_2} - 1) + q_{CH_4}(Le_{CH_4} - 1)}{q}$$
(6)

where the parameter q_i (i refers to H₂ and CH₄) is the nondimensional heat release associated with the consumption of species i, defined as

$$q_{i} = \frac{QY_{i}}{c_{p}T_{u}}$$
⁽⁷⁾

Here, Q is the heat of reaction, c_p is the unburned gas specific heat, T_u is the unburned gas temperature and Y_i is the supply mass fraction of species i. The quantity $q = q_{H_2} + q_{CH_4}$ is the total heat release and Le_{H_2} and Le_{CH_4} are the Lewis numbers of hydrogen–air mixture at ϕ_{H_2} and methane–air mixture at ϕ_{CH_4} , respectively.

3. Laminar burning velocity and Markstein length

For an outwardly propagating spherical flame, the stretched flame velocity, S_n , reflecting the flame propagation speed, is derived from the flame radius versus time [25]



Fig. 2 - Schematic of constant volume combustion vessel.



Fig. 3 – Methane based ϕ_{CH_4} and hydrogen based ϕ_{H_2} equivalence ratios at different hydrogen fractions.

$$S_n = \frac{dr_u}{dt} \tag{8}$$

where r_u is the radius of flame in Schlieren photograph and t is the elapsing time from spark ignition.

A general definition of stretch at any point on the flame surface is the Lagrangian time derivative of the logarithm of the area A of any infinitesimal element of the surface

$$\alpha = \frac{d(\ln A)}{dt} = \frac{1}{A} \frac{dA}{dt}$$
(9)

For the outwardly propagating spherical flame, the flame stretch rate can be deduced in the following form

$$\alpha = \frac{1}{A}\frac{\mathrm{d}A}{\mathrm{d}t} = \frac{2}{r_{\mathrm{u}}}\frac{\mathrm{d}r_{\mathrm{u}}}{\mathrm{d}t} = \frac{2}{r_{\mathrm{u}}}S_n \tag{10}$$

In respect to the early stage of flame expansion, there exists a linear relationship between the flame speed and the flame stretch rate [18].

$$S_l - S_n = L_b \alpha \tag{11}$$

where S_l is the unstretched flame propagation speed, obtained as the intercept value of S_n at $\alpha = 0$ in the plot of S_n against α . Burned gas Markstein length L_b is the negative value of the slope of S_n - α curve.

The characteristics of the igniter can influence the measured value of burning velocity. Previous study showed that flame speeds were independent of ignition energy when flame radius was larger than 5 mm. This phenomenon was also observed by Bradley et al. [18], Lamoureux et al. [25] and Huang et al. [30]. To avoid the effect of ignition energy and pressure rise in the combustion chamber, the flame photos in the range of 5–25 mm were used in the analysis. Besides, because Eqs. (8)–(11) were used only for the smooth flame front, the measurements were also restricted before the occurrence of the cellular structure [35].

In the early stage of flame propagation, the flame undergoes an isobaric developing process, the unstretched laminar burning velocity, $u_l,$ is related to $S_l\ from\ mass conservation\ across the flame\ front$

$$A\rho_{\rm u}u_{\rm l} = A\rho_{\rm b}S_{\rm l} \tag{12}$$

where A is the flame front area, ρ_u and ρ_b are the unburned and burned gas densities, respectively. The unstretched laminar burning velocity, u_l , can be obtained from Eq. (12)



Fig. 4 – Flame propagation speed versus flame stretch rate at different initial pressures and initial temperatures for methane-hydrogen-air mixtures.



Fig. 5 – Unstretched flame propagation speed and Markstein length at different initial pressures and initial temperatures for methane-hydrogen-air mixtures.

$$u_{l} = \rho_{b} S_{l} / \rho_{u} \tag{13}$$

In this study, the laminar flame thickness, δ_{l} , is determined as [34,36]

$$\delta_{l} = \frac{D_{th}}{u_{l}},\tag{14}$$

where $\ensuremath{\mathcal{D}_{th}}$ is thermal diffusivity of unburned gas, which is defined as

$$D_{\rm th} = \frac{\lambda}{\rho_{\rm u} C_{\rm p}} \tag{15}$$

Here, λ and c_p are the thermal conductivity and specific heat of unburned gas, respectively.

4. Results and discussions

4.1. Flame propagation velocity and Markstein length

Fig. 4 shows the flame propagation speed versus flame stretch rate at an equivalence ratio of 0.8. In the early stage of flame propagation where the flame radius is small, the stretch rate of flame front surface is large. Removing the data affected by the ignition energy and electrodes during the early stage of flame development and large radius where pressure is increased, a linear correlation between the stretched flame propagation speed and the flame stretch rate is demonstrated. The unstretched flame propagation speed, S_l, is obtained as the intercept value of S_n at $\alpha = 0$ in the plot of S_n versus α . Importantly, this gives the value of laminar burning velocity u_l from Eq. (13). The flame propagation speed versus flame stretch rate for hydrogen percentage of 60% at initial temperature of 373 K and at different initial pressures is illustrated in Fig. 4a. The stretched flame propagation speeds decrease with the increase of stretch rate at all initial pressure conditions, and the gradients of $S_n-\alpha$ lines take the negative values, representing the positive values of Markstein length, L_b. Fig. 4b shows the flame propagation speed versus flame stretch rate at $P_{\rm u}=0.5\,MPa$ and $X_{\rm H_2}=20\%$ under different initial temperatures. Increasing the initial temperature from 303 K to 443 K shows little influence on flame stretch, but flame propagation speed increases with the increase of initial temperature. Fig. 4c shows the $S_n-\alpha$ curve for the mixtures with different hydrogen fractions at $P_{\rm u} = 0.5$ MPa and $T_u = 373$ K. The results show that the stretched flame propagation speeds are increased and the burned gas Markstein lengths (L_b) are decreased with the increase of hydrogen



Fig. 6 – Schlieren images of methane-hydrogen-air mixtures at different initial pressures (X_{H2} = 60%, T_u = 373 K).

fraction, and this indicates that the flame instability will be increased with the increase of hydrogen fraction.

Fig. 5 shows the unstretched flame propagation speed (S_1) and Markstein length (L_b) versus initial pressure and hydrogen fraction at an equivalence ratio of 0.8. As shown in Fig. 5a, the unstretched flame propagation speed is decreased with the increase of initial pressure. For a fixed initial pressure, the unstretched flame propagation speed is increased with the increase of hydrogen fraction. Variations of Markstein length (L_b) with initial pressure are illustrated in Fig. 5b. The results show that *L*_b is decreased with the increase of initial pressure, and this indicates the increase of flame front instability at the elevated pressure. Fig. 5c shows the S1 versus hydrogen fraction at different initial temperatures. S₁ shows an exponent increasing trend with the increase of X_{H2}. For a given hydrogen fraction, S₁ is increased with the increase of initial temperature. Fig. 5d shows that L_b decreased remarkably with the increase of X_{H_2} and reveals the increase of flame front instability when hydrogen is added. With the increase of hydrogen fraction, the flame behavior tends to be similar to that of lean hydrogen flame, which is diffusional-thermally unstable. The study shows no appreciable difference in L_b at different initial temperatures.

4.2. Flame stability and cellular structure

Cellular flames tend to develop in the mixtures in which the deficient reactant constituent is also the constituent with the largest diffusivity. This kind of flame cellularity was experimentally observed for both the plane flame from the study of Markstein [37] and the spherically expanding flame from Manton et al. [38]. The formation of cells in such cases is a consequence of diffusional-thermal instability, which results from the competing effects of heat conduction from the flame and reactant diffusion toward the flame [39,40]. Lewis number (*Le*) is used to represent the instability and



Fig. 7 – Schlieren images of methane-hydrogen-air mixtures at different initial temperatures (X_{H_2} = 40%, P_u = 0.5 MPa).

Lewis number is defined as the ratio of heat diffusivity of the mixture to mass diffusivity of the limiting reactant. When Le is below some critical value, Le*(slightly lower than unity), the diffusional-thermal instability can be observed during the initial phase of propagation, i.e., the flame radius is the order of flame thickness [40]. On the other hand, mixtures with Le larger than unity are stable due to the diffusional-thermal mechanism. However, some experiments also observed the cellular flames in such cases [41-43]. This is regarded as the intrinsic Darrieus-Landau hydrodynamic instability resulting from the interaction of the flame with the hydrodynamic disturbances [40], and this instability is enhanced when the thermal expansion ratio (σ), which is the ratio of unburned gas to burned gas at two sides of the flame front, is increased and the flame thickness (δ_l) is decreased [20,27]. At the early stage of flame development, instability of flame is mainly influenced by thermo-diffusive factor. However, with flame development and flame radius increasing, the hydrodynamic factor becomes the dominant factor [27]. Initially, the cellular instability is suppressed by the strong curvature associated with small flame radii. As the flame expands and stretch lessens, there appears a critical radius (R_{cr}) where the cellular instability can no longer be suppressed, resulting in the instantaneous formation of cells over the entire flame surface. The transition position depends on the combined influence of the hydrodynamic and thermal-diffusive instabilities. Bechtold and Matalon [40] and Addabbo et al. [44] presented an explicit expression for the onset of cellularity by introducing a critical Peclet number (Pecr), which is the value of flame radius at transition or critical radius (R_{cr}), normalized by laminar flame thickness (δ_l).

Fig. 6 shows the Schlieren images of the expanding spherical flame at $T_u = 373 \text{ K}$ and at different pressures for methane-hydrogen-air mixtures. The flame front remains a smooth surface at an initial pressure of 0.1 MPa. At the elevated pressures, the flame surface first appears to be smooth, then some cracks grow and branch until eventually cellular structure appears over the entire flame surface. The onset of cellularity occurs at early position with the increase of initial pressure. The effective Lewis number (Leeff) keeps the same value at different initial pressures for the fixed ϕ and X_{H_2} , as shown on the right side of Fig. 6, and this reveals little variation in diffusional-thermal instability with the increase of initial pressure. The hydrodynamic instability originated from gas thermal expansion (σ) and flame thickness (δ_l). The study shows that the thermal expansion ratio remains the same at different initial pressures, and the remaining parameter that governs the hydrodynamic instability is the flame thickness. Fig. 6 also lists the flame thickness, seen in the table attached on the right side of the figure. The increase in hydrodynamic instability with increasing initial pressure resulted from the decrease in flame thickness. The combined result of the two instabilities leads to the flame instability enhanced with the increase of initial pressure.

To identify the effect of initial temperature on the flame instability, the Schlieren images at different initial temperatures are provided in Fig. 7. The results show that flame instability is insensitive to the initial temperature. When initial temperature is increased, the flame instability shows little variations. The effective Lewis number (Le_{eff}) decreases slightly with the increase of initial temperature for the fixed ϕ and X_{H_2} , as shown on the right side of Fig. 7, and this



Fig. 8 – Schlieren pictures of methane-hydrogen-air mixtures at $P_u = 0.5$ MPa, $T_u = 373$ K and at various hydrogen fractions.

indicates that initial temperature has little influence on the diffusional-thermal instability. The values of thermal expansion (σ) and flame thickness (δ_1) at different initial temperatures are also provided on the right side of Fig. 7. With the increase of initial temperature, both σ and δ_1 are decreased. The factor σ leads to the decrease of hydrodynamic instability while factor δ_1 leads to the increase of hydrodynamic instability. The combined influence of the two factors results in little variation in flame front instability at different temperatures.

Fig. 8 shows the Schlieren images at different hydrogen fractions under the equivalence ratio of 0.8. No instabilities are observed for the methane–air flame ($X_{H_2} = 0\%$), although some cracks appear on the flame front surface, which are caused by the ignition disturbance. However, these cracks do not develop during the flame propagation. When hydrogen is added, flame instability can be observed, especially in large $X_{\rm H_2}$ cases. Earlier onset of cellular flame structure is presented at $X_{\rm H_2}=60\%$ compared to $X_{\rm H_2}=40\%.$ With the increase of X_{H_2} , the effective Lewis number is decreased, indicating the increase of the diffusional-thermal instability. Flame becomes more similar to that of lean hydrogen-air flame at large hydrogen fraction, which is diffusional-thermally unstable [45]. In addition, the thermal expansion ratio gives little variation at different hydrogen fractions while the flame thickness is decreased significantly with the increase of hydrogen fraction, as shown in data on the right side of Fig. 8. Advance of the onset of cellular instability is from the thermal-diffusional instability as decreased Lewis number and the hydrodynamic instability as decreased laminar flame thickness. For methane–hydrogen–air flame, when $X_{\rm H_2}$ becomes large, both the diffusional-thermal instability and the hydrodynamic instability are enhanced, leading to the advancing onset of cellular flame.

Fig. 9 shows the stretched flame propagation speed versus flame radius at different initial pressures and hydrogen fractions. In the early stage of flame propagation, the value of S_n is decreased as the flame propagated and the Schlieren flame photos showed sign of cellularity. However, at certain positions as indicated by the asterisks, the flame front becomes cellular pattern and from this position S_n starts to increase remarkably due to the increase of the flame front area [36,46]. The flame radius of this position is defined as the critical radius, R_{cr} . When normalized by flame thickness, δ_{l} , it gives the critical Peclet number, Pe_{cr} , for the given conditions. Clearly, the value of R_{cr} is decreased when initial pressure and/or hydrogen fraction is increased.

To clearly illustrate the effect of cellular structure on flame propagation speed, Fig. 10 plots the stretched flame speed against the stretch rate. There is a rapid increase in S_n at critical flame radii. This transition point represents the onset of the cellular structure [32]. The information revealed from S_n versus α curve is consistent with the photo observed.



Fig. 9 – Stretched flame propagation speed versus flame radius.



Fig. 10 – Stretched flame propagation speed versus stretch rate.

Fig. 11 shows the critical radius (R_{cr}) and critical Peclet number (Pe_{cr}) versus hydrogen fraction for methanehydrogen-air mixtures. When X_{H_2} is less than 20%, no cellular instability is observed within the observed flame radius range. As X_{H_2} increases, both R_{cr} and Pe_{cr} are decreasing. For a fixed X_{H_2} , R_{cr} is decreased when initial pressure is increased. This indicates that the destabilizing effect will become stronger when increasing initial pressure and hydrogen fraction.

4.3. Laminar burning velocity

Fig. 12 shows the unstretched laminar burning velocity at different initial pressures and initial temperatures for lean methane-hydrogen-air flames. The unstretched laminar burning velocity is decreased monotonically with the increase of initial pressure and is increased with the increase of initial temperature. As the pressure is increased, the intensity of the temperature-sensitive, two body, branching reaction: $H + O_2 \rightarrow OH + O$ is approximately fixed due to the insensitivity of adiabatic temperature to the increase in pressure, while the three body, temperature-insensitive, inhibiting reaction: $H + O_2 + M \rightarrow HO_2 + M$ is enhanced, and a retarding effect is therefore imposed on the overall progress of the reaction with increasing pressure [19,47,48]. An



Fig. 11 – Critical radius and Peclet number versus $X_{\rm H_2}$ at different initial pressures and at different hydrogen fractions.



Fig. 12 – Laminar burning velocity (u_l) at different initial pressures and initial temperatures for methane–hydrogen– air mixtures.

increase in the upstream temperature leads to the increase of the adiabatic temperature, hence the reaction rate, and the dependence is more sensitive at high T_u because of the Arrhenius factor. u_l is increased with the increase of X_{H_2} , and the increasing trend becomes more obviously at larger X_{H_2} . Hydrogen addition increases the concentration of radicals from the activation reaction region where they are generated [13,49]. In addition, the volumetric heating value of hydrogen is lower than that of methane, but the fuel-air ratio of hydrogen is four times as that of methane at the stoichiometric condition. Thus, increasing hydrogen fraction decreases the amount of air in which nitrogen is a major component in determining the specific heat of the mixture, consequently resulting in the increase of adiabatic temperature and enhancing the reaction rate with the increase of hydrogen fraction.

5. Conclusions

An experimental study on laminar burning velocities and onset of cellular instabilities of methane-hydrogen-air mixtures using the spherically expanding flames was conducted at different initial pressures, initial temperatures, and hydrogen fractions. The conclusions are summarized as follows.

- (1) The unstretched flame propagation speed and the unstretched laminar burning velocity increase with the increase of initial temperature and hydrogen fraction, and they decrease with the increase of initial pressure.
- (2) With the increase of initial pressure, advancement of onset of cellular instability is presented and the critical radius and Markstein length are decreased, indicating the increase of hydrodynamic instability with the increase of initial pressure. The initial temperature is insensitive to the flame instability.
- (3) At the equivalence ratio of 0.8, significant decrease of critical radius and Markstein length with the increase of hydrogen fraction are presented, indicating the increase in both diffusional-thermal and hydrodynamic instabilities when hydrogen fraction is increased.

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